Airc RSI	raft serial no. JK/CVLN	serial no.Aircraft Long Term Storage and Return to ServiceCVLNWorksheet (Cavalon)			Aircraft registra	ation no: (e:	3-
		Unique worksheet no.	(if required/used):		Worksheet type: LTSRS		
Task No Task Description Repetition or comments				Actions taken &	comment	Eng'r initial	A3-7 Certifier
Purp of st Refe Mos	Purpose of this worksheet: To be applied when preparing for storage, when in storage, or when returning the aircraft to service from a long period of storage/disuse, normally longer than one year. Also to be used when the storage of an aircraft under Form F178 exceeds one year. Refer also to Maintenance Manual RSUK0288. Most of the checks and serviceability are 'on condition', meaning the Engineer has the responsibility to decide if it is acceptable for service.					ng period ice.	
Pre	paring for sto	rage task list					
NB:	It is assumed that t be the case (e.g we	he aircraft will be stored in t floor, condensation, signi	່a clean, dry, well-ventilated (but not ne ficant dust) then RSUK should be consເ	cessarily heated) ulted to consider v	building with a se whether additiona	aled floor. S I actions are	Should this required.
P1	Drain fuel		Do not store for subsequent replacement. Mogas should not be used after 3months				
P2	Arrange 240V AC r battery charger	nains supply for Cetek	Battery will be charged via the external charging point as noted in S5 below.				
P3	Prepare engine in a recommendations	accordance with Rotax	Ref: Rotax Heavy Maintenance Manual Section 71-00-00 para 5.2.				
P4	Protect aircraft enc aircraft cover	losure/engine with RSUK					
P5	Remove rotors from suitable crate with	n aircraft and store in a soft liner and interleaving	Available from RSUK				
P6	Wrap rotor head in	a dust-sheet					
P7	Fit cover to pitot-tu	be (with vent for breathing)	Must have flight-safety lanyard or be attached to tie-down cord				
P8	Cover static vents of porous adhesive ta	2) with a piece of micro- pe	Each tape must be attached to a lanyard of conspicuous colour				
	Intentionally blank		Intentionally blank	Intentionally blan	k	Intentionally blank	Intentionally blank

Aircraft serial no. RSUK/CVLN Unique worksheet no		Aircraft Long T V Unique worksheet no.	erm Storage and Return t Vorksheet (Cavalon) (if required/used):	to Service	Aircraft registra Worksheet dat Worksheet typ	ation no: e: e: LTSRS	G-
Task No	Task	Description	Repetition or comments	Actions taken &	comment	Eng'r initial	A3-7 Certifier
In-s	storage task li	st – the 3month int	erval				•
S1	Jack aircraft, spin v pressures and tyres	vheels. Check tyre s for cracks	Spinning avoids flats and brake binding No cracks will be acceptable for Return to Service	Record each action initial on a separator to this worksheet	on, its date and ate sheet attached		
S2	Check engine for co shaft/flange, conne	orrosion (propeller ctors)	Clean and protect with WD40 or if required	Record each action initial on a separator to this worksheet	on, its date and ate sheet attached		
S3	Check for oil or coc	olant leaks	Arrange rectification if found.	Record each action initial on a separator to this worksheet	on, its date and ate sheet attached		
S4	Check for bird or ro droppings	odent nests, wash-off any	Air filters, exhaust, behind instrument panel (use mirror and torch), enclosure foot-wells, luggage lockers, engine bay. Pull-back stick gaiters for access to controls.	Record each action initial on a separation to this worksheet	on, its date and ate sheet attached		
S5	Check the open-cir battery. This may b special magnetic ch aircraft's external p lower engine cowlir connector and usin connector terminals external load on the is taken.	cuit voltage (OCV) of the e done by attaching the harging cord to the ower point (on the RH ng), separating its in-line g a multimeter at the two s. There must be no e battery when the reading	If less than 12.6VDC (i.e. 2.10 volts per cell) then charge for a 12 hour period or until this OCV is reached. Use only the Cetek charger or the battery may be damaged	Record each action initial on a separa to this worksheet	on, its date and ate sheet attached		
S6	Periodically clean a Inspect rotors for co RSUK	aircraft including rotors. orrosion, if found consult	Do not use washing-up liquid	Record each action initial on a separate to this worksheet	on, its date and ate sheet attached		
S7	Annually repeat the accordance with Ro	e engine preparation in otax recommendations	Ref: Rotax Heavy Maintenance Manual Section 71-00-00 para 5.2 & 5.3	Record each action initial on a separator to this worksheet	on, its date and ate sheet attached		

Ret	Return to Service task list					
	Airframe Inspection	All items – repeat inspections as shown unless stated otherwise				
1	Remove upper and lower engine cowlings, mast cowling. Thoroughly check aircraft for evidence of missing parts or instruments.	Check against aircraft SAC that aircraft is still to the required build standard. Refer to AMM RSUK0288 for detail information on cowling removal and replacement				
2	Check security of upper mast attachment	Whilst standing on the rear seat rock the upper mast independently in the pitch and roll axes and confirm no free- play evident in the two rubber bushes or their mountings.				
3	Check - Bolt security – other					
4	Inspect - Wheel bearings smooth operation (3 places)	Wheel bearings sealed for life. Raise aircraft with jack under rear keel.				
5	Op/C - nosewheel fork for straightness and free operation.	Tip the aircraft onto its tail and visually assess the straightness of nose-wheel fork. The nose-wheel fork must rotate freely to the limit stops in the nose of the aircraft. There should be minimal play in the bearings of the nose-wheel fork.				
6	Inspect nose-wheel fork rubber buffer for security and condition	No cracks or splits				
7	Inspect - landing gear spar and attachments to airframe for damage or fatigue (cracks & deformation).	If any cracks or deformation found then ground aircraft and contact RSUK immediately				
8	Inspect – tyres for wear or damage. Replace if needed.	No fabric to show through the tread area. Recommended 0.5mm min tread. Ensure no flat spots or wall cracks from storage				
9	Check - tyre pressures & tyre creep (mainwheels 1,8-2.2bar if heavily loaded, nose 2.0-2.2bar)		Pressures OK Nose			
			Main LH	Main RH		

10	Change brake fluid	Recommended at 3years, or when brakes become spongy. Refill from master cylinder with callipers immersed in fluid. If system is spongy after bleeding, check discs for flatness and wheel bolts for straightness.	(on condition)	
11	Inspect - airframe for damage, twisting, buckling, or other deformation, or cracks, especially at welded joints at bottom of the mast.	If found ground aircraft and call RSUK for advice. Use of crack detection fluid at base of mast is appropriate to ensure a thorough check is done.		
12	Inspect - External structure of enclosure sound and firmly fixed to airframe			
13	Inspect – Canopy attachment to enclosure, canopy latch forces and function of canopy latch interlock	Refer to Pilots Handbook RSUK0287 Section 6.12 for detailed procedures		
	Electrical/instruments			
14	Inspect - panel connections for security			
15	Inspect – sealed battery for leakage	Ensure battery is charged and holding charge (use Cetek charger for 12 hour period or follow S5 above).		
16	Op/C Check strobe function if fitted			
17	Op/C check nav light function if fitted			
18	Op/C check backup fuel pump functions			
19	Op/C check landing light function if fitted			
20	Op/C – Roll trim. Operate roll trim (where fitted) fully left. Ensure panel indicator shows fully left. Then operate trim fully right. Ensure indicator shows fully right			

	Rotor head			
21	Time-related 1500hr: Renew main bearing	Replace bearing at 1500hrs (no extension permitted). Bearing bolt torque 160Nm+/-20Nm (plus split pin) NOTE: when tightening hub onto backing plate ensure that the clearance between the main gear and bendix gear is minimised from 0.05 to 0.15mm Glue bearing temp sensor in with hot melt adhesive. Clearance of rotor speed sensor to gear is 1 to 2mm (confirm function via tacho)	1 st inspection Name: A3-7 authorisation no: Sig: 2 nd inspection Name: A3-7 authorisation no: Or, qualified pilot licence no: Sig:	
22	Check split pin present and no sign of chaffing or looseness. If present, check nut torque and replace split pin.	Second signature required if pin replaced	1 st inspection Name: A3-7 authorisation no: Sig: 2 nd inspection Name: A3-7 authorisation no: Or, qualified pilot licence no: Sig:	
23	Op/C - Ring gear security and bolt attachment	Note any wear patterns		

24	Check, Service/lube - teeter bolt & bearings for damage & wear.	Regrease via nipple on top of rotor (where fitted). Grease with Castrol LM or equivalent . Removal, clean, inspect and refit is recommended every 100hrs. If wear or signs of distress, remove rotor assembly, inspect and replace bushes or bolt if required. Clean, regrease & refit. NB: Excess wear is more than 0.5mm of vertical play, bolt to bushes, and will cause rotor vibration. Nut must not be more than finger tight, about 1 to 2Nm, and the bolt able to	1 st inspection Name: A3-7 authorisation no: Sig: 2 nd inspection Name: A3-7 authorisation no: Or, qualified pilot licence no: Sig:	
25	Check bushes in teeter-tower sides. If worn,	turn by hand. Small sideways float between hub bar		
26	Time related 100hr: Service/lube –gimbal joints, check for wear & regrease.	Grease with Castrol LM or equivalent. If wear evident or noticeable looseness, disassemble gimbal joints, check for wear, regrease and reassemble. Torque up bolts to clamp side plates to gimbal block as per RSUK0288 Strip and inspect recommended every 200hrs of operation Check - with the rotor removed assess the stick force/stiction effect when moving aft and fwds. The stick should be moved slowly away from each end- stop until the "stiction" effect is overcome. The maximum allowable forces are in RSUK0288. NB: the difference in force is due to the weight of the rotor-head acting on the control cable.	Record: Stick force moving aft Stick force moving fwd 1 st inspection Name: A3-7 authorisation no: Sig: 2 nd inspection Name: A3-7 authorisation no: Or, qualified pilot licence no: Sig:	
27	Check four split pins present and secure	Main bearing, teeter bolt, pitch and roll bolts. Required even if no disassembly		
		actions.		
28	Lubricate Bendix gear & spiral gear	WD40 or similar		
29	Lubricate rotor brake pivot.	WD40 or similar		

30	Inspect - brake pad for function & wear	Change pad and backplate as one unit		
21	On/C Chack Trim cylinder for free function	(Service item)		
51	and shaft damage or excess seal leakage.	Seal service kit is available from KSOK		
32	Op/C – Check Roll Trim cylinder (if fitted) for			
	free function and shaft damage or excess seal			
	leakage.			
33	Protect bare metal with Motor Plus, WD40,			
	chain wax or equivalent			
	Rotor Head Controls			
34	Service/lube - clean rod ends (if appropriate)			
35	F/C- rotor head reaches pitch and roll stops			
36	Inspect - all cables undamaged, all bearings	Pull-back stick gaiters for access to		
	free, all bearing retaining rivets secure, no	controls. Check also for insect or animal		
	foreign bodies or debris in control tubes	residue		
37	Op/C - for free play in stick control e.g.			
	bearings or wear			
	Rudder controls			
38	Op/C - Check pedals for ease of movement			
39	Inspect for cable freedom of movement at tail			
	and pedal attachment, and turnbuckle			
	wirelocking.			
	Check Nicopress sleeves for signs of			
40	movement.	Destinutes attention to achie quit from		
40	Inspect - rudder cables for frays, corrosion,	Particular attention to cable exit from		
4.4	wear or challing	Keel-lubes.		
41	Inspect - tail bearings for looseness and			
40	Increation of operation	Lastitud if lassa remove and refit with		
42	15Nm)	loctite 243. Check to 12Nm		
43	Inspect tail and rudder for signs of composite	Include waggling the side fin in case of		
	damage.	internal structural damage.		
44	Inspect – rod-ends and plate at base of rudder			
	for free rotation, security & wear			
45	Inspect – rudder to tail fastenings	Check to 12Nm	Confirm if possible rudder offset to	
			pedals	
46	Inspect – security of rudder trim tab			
47	F/C - rudder control cable tension	For limits and methods see AMM RSUK0288 Section 9 I)	Gauge no. Reading	

48	Inspect – that all rod end joints are fitted with a snubbing failsafe washer.			
49	Check that all control system bolts are correct items, properly fitted and tight			
50	Engine NOTE! All engine checks to be in accordance with manufacturers manual!	For engine servicing refer to the engine manual issued with the aircraft (Rotax 912ULS or 914UL). The full annual engine service is required only when no engine servicing has been carried out in the last 12 months. Otherwise apply 'on condition'. Servicing must be carried out in line with, and recorded on, the Rotax service schedule contained within the 'Line Maintenance' manual for the engine fitted. The Rotax service centre will advise additional checks subject to the method of storage used. (e.g. borescope checks)		
51	Wirelocking – ensure present on: Oil tank drain plug, Aftermuffler (transverse types), Oil banjo under engine, Carb air filters (if wire-locked), Oil pump			
52	Engine service fasteners	If the magnetic inspection plug or the crankshaft locking screw plug are disturbed then any wire-locking present must be properly reinstated		
53	Inspect – oil tank breather pipe for blockage			
54	Service/lube - Lubricate carburettor choke levers if no free movement	HSC2000 spray grease or equivalent		
55	Service/lube - Ensure choke and throttles move freely from stop to stop, and that turbo detent can be felt correctly. Ensure cables are synchronised.			
56	Inspect – engine mount rubbers for deterioration			
57	Inspect engine bearer bolts for paint stripe, and if moved, re loctite and tighten to 35Nm. Otherwise check bolt torque. Re-apply paint stripe as required.			
	Fuel system			

58	Check - whilst fuel tank(s) empty, check that low fuel warning LED lights. Service/lube –Fuel tanks. Flush tanks with about 1 litre of fuel then fill with fresh. Ensure water drain points function correctly on refill, and confirm no tank debris. Check – when fuel tanks filled check that low fuel warning light extinguished	There may be a small amount of leakage until the rubber seals swell due to the effect of the fuel. If the fuel drain wirelock is removed, it MUST be replaced, with a dual inspection signature.	1 st inspection Name: A3-7 authorisation no: Sig: 2 nd inspection Name: A3-7 authorisation no: Or, qualified pilot licence no: Sig:	
59	Service/lube - Change fuel filters if dirty (two filters on 912ULS engine, two on 914UL)	Plastic and metal filters are used in different locations – ensure correct replacements are used		
60	Inspect - fuel tank cap for seal deterioration & security of fit, and cleanliness of vent hole.			
61	Inspect – security of fuel tanks and tightness of tank straps	Fuel tanks must not be deformed by straps and protective strips must be in place.		
62	Op/C - functionality of fuel gauge	ie that the reading matches that shown on the tank sight gauge.		
63	Inspect – fuel-tank breather pipe for blockage.	If 914UL engine also inspect clear airbox/carb-tray drain pipe		
64	Inspect - all hoses for cracks and deterioration in the visible areas adjacent to the barbed metal fittings.	Change as required		
	Pre rotator			
65	Op/C – whilst turning the uj located at the base of the mast by hand (thru a full rotation) – check drive shaft joints for free movement and bearings for play etc			
66	Inspect - universal joints for corrosion	Clean as required (use a kitchen plastic scouring pad) and spray with oil or chain wax		
67	Inspect - drive unit (bendix) engagement to rotor drive gear.	Do not grease this unit! – very light oil only or it will start to jam.		
68	Inspect - Ensure slider shafts move freely, and are greased			
69	Not used			

	Trim System, Rotor Brake & Pneumatics			
70	Inspect – all hoses for leaks and slave cylinder(s) for looseness			
71	Op/C – Roll trim. Operate roll trim (where fitted) fully left. Ensure panel indicator shows fully left. Then operate trim fully right. Ensure indicator shows fully right			
72	Change (or dry out) compressor water absorber	Normal 100hr task		
73	Inspect – compressor. Listen for undue noises in operation.			
74	Op/C - Full functional check pneumatic system - refer as required to the maintenance manual for fault finding and rectification, and a more comprehensive understanding of the test background.	With selector set to 'Brake' position, engage brake by pressing button, confirm operation, and that function is acceptable. Pressurise to maximum. Change to flight – check 8 sec max to release air from brake system). In 'Flight' position check that trim goes on and off in same direction as button (inc left stick if fitted). In 'Flight' position, stick forward. Start pre rotator. Ensure bendix drive cylinder rises to engage, and when the stick is pulled back it disengages. Stick to front, release pre rotator and confirm that pressure is applied to trim and stick comes back slightly. In 'Brake' position, put 3 bar pressure on and ensure pre rotator does not function. Press the 'Interlock release button' and ensure that pre rotator functions with brake engaged.		
75	Op/C – check compressor can give full		Note pressure obtained	
	pressure of 7bar (~8bar with new compressor). If under 5.5bar, either find leak or replace.			
	HTC Propeller			
76	F/C - tracking to manufacturers recommendations	(none required at the time of writing)		

				1
77	Check - prop bolt torque stripe between bolt thread and propeller hub has not been broken (indicating that the bolt has slackened).	If torque stripe broken or missing, remove bolts, inspect, and refit with loctite 243 – and re-apply torque stripe (Engineer task!) Removal of spinner (if fitted) will be required.		
78	Measure prop blade pitch angle relative to hub.	Recommend pitch to be within 0.5deg of each other.	Blade 1	
	20.5deg (914UL)		Blade 2	
			Blade 3	
			Hub (datum) angle	
79	Inspect - blades to manufacturers recommendations for any damage, splits etc.	Repair only as manufacturer's recommendations (see AMM)		
	Rotors			
80	Inspect - blades to manufacturers	Repair only as manufacturer's		
	recommendations for any damage, splits etc.	recommendations		
81	Inspect - blade to hub bar attachment bolts for	Light corrosion should be coated in		
	corrosion	chain wax or wD40/equivalent. If bolts		
		replace as appropriate. Lubricate with		
		chain wax or equivalent on refitment		
82	Check - torques on blade to hub bar bolts/nuts	If any evidence of blade to hub looseness, disassemble blades from hub bar. Check holes for wear or fretting Bolt torque 25Nm. Refer to Sect 9 General Notes of the Maintenance Manual for re-usage of nyloc nuts.		
	Other			
83	Inspect - for wheel-brake pad wear. Replace as necessary, and if less than 2mm pad remaining (later pads have witness groove).	If calipers are sticking or uneven wear is found, loosen/turn wheel bolts and check for straightness – if OK retighten. Alternatively, clean brake pad bushes & lubricate calipers around seal		

84	Inspect – brake ratchet pawl for excessive wear. If found, replace.	Teeth of lever must not be visibly deformed or protrude less than 1.5mm		
85	Remove pitot and 2-off static vent covers			
86	Inspect - Confirm all placards readable and in line with Operating Limitations	See Pilots handbooks for placards required – or consult CAA TADs publication.		
87	Check aircraft weight and balance	No annual check required, but confirm weighing certificate available and matches wt on placard		
88	Inspect all seat belt attachment points for tightness and security			
89	Inspect each seat belt for damage or frays, and for security of main connection			
90	F/C - ASI function	Verify function using field-test kit RSD7179 (as described in AMM RSUK0288 Section 9)		
91	F/C – compass function	Position aircraft on each cardinal point and confirm compass readings consistent		
92	F/C – altimeter function	Verify function using field-test kit RSD7179/RSD7180 (as described in AMM RSUK0288 Section 9)		
93	Op/C - Instrument checks	Transponder - Check that mode S code matches G-INFO database. Check that pressure altitude consistent with altimeter reading. Full functional check highly recommended. Radio – confirm PTT buttons cause 'T' on panel. Turbo TCU data (where fitted) may be downloaded for analysis		
	Final ground run checks prior to release			
94	Inspect - Power plant and coolant system for leaks			
95	Inspect – security of oil-thermostat insulator pad		 	
96	Inspect – instruments for measurements consistent with ambient conditions			
97	Replace all cowlings and check all access covers secure			

98	Op/C - verify correct function of Fire-Warning system	Turn on Master switch. The fire warning lamp will pulse red three times to confirm correct system function and then go off if the system functions normally. If not it will stay solid red (system fault), in which case stop and investigate				
99	Securely tie aircraft down and run to full power. Ensure engine rpm achieves at least 5,400 on one fuel pump only, and with both pumps running		RPM achieved:			
10 0	Complete mag drop checks at 4,000rpm	See Pilots Handbook for limits	Mag drop#1			
			Mag drop#2			
10 1	Confirm 'Gen' light is on when engine not running, and off (or flickering gently) when running at above 2000rpm.					
10 2	Confirm low fuel lamp is not lit (providing the fuel covers the sensor)					
10	Ensure all log book entries completed					
3	appropriately, and service record up to date					
Cont	irm Service bulletins incorporated (from RSUK w					
Cont	irm Rotax Service Bulletins incorporated (from R					
Cont Up-te	irm Mandatory Permit Directives incorporated (fro p-date information must be checked!					
CAP 747 Document date or issue checked, plus notes:						
EAS	A MPD or AD check (EASA website): note date of					
Cont Note	irm compliance to BG06 Type Approval Data Sho any non-compliances and actions taken.					

Tasks completed by (name):					
Signature:	Initial:	Airframe bours logged:			
Date:	(to compare to				
	check sheet)	Aircraft hourmeter hrs logged:			
		-			
Permit Maintenance Release: The work recorded above (all p been completed to my satisfaction and in that respect the air considered fit for flight.	Comments:				
Signature:	Initial:				
Date:	(to compare to check sheet)				
Inspector or licence no.: Company Approval ref					
Inspector Authority: CAA letter ref 9/ dated					
Note to Engineer; remember to reference this worksheet and RSUK0288 within the logbooks, together with your CAA authorisation code. Work undertaken may be noted on this worksheet, or if required on another sheet (such as F093) also referenced in the logbook. Modifications undertaken must be noted with their MC approval no. Check the back pages to complete these too for modifications, service bulletins, MPDs, etc.					

Appendix

Requirements for certifying signatures/initials on this worksheet

With the exception of "Permitted Pilot Maintenance" (see the relevant RSUK Aircraft Maintenance Manual and CAA publication CAP 733), all maintenance work on RSUK gyroplanes must be certified by a CAA A3-7 Authorised Person.

Case 1: for work not involving engine controls, or flying controls, or vital structural points

The person(s) performing the work should complete the worksheet columns as below:

- If the person completing "Eng'r" does not have A3-7 authorisation there must be a second initial by an A3-7 authorised person in each adjacent "A3-7 certifier" cell, denoting acceptance of the task specified.
- If the person has A3-7 authorisation the "Eng'r" cell should be struck out and a single entry of initials made in the A3-7 certifier cell

Case 2: for work where engine controls, or flying controls, or vital structural points are disturbed, where a duplicate inspection is required (and shown in the worksheet).

The person(s) performing the work should complete the worksheet columns as shown above and repeated below:

- If the person completing "Eng'r" does not have A3-7 authorisation there must be a second initial by an A3-7 authorised person in each adjacent "A3-7 certifier" cell, denoting acceptance of the task specified.
- If the person has A3-7 authorisation the "Eng'r" cell should be struck out and a single entry of initials made in the A3-7 certifier cell

In addition to the above there is a requirement for inspection, then duplicate inspection (by an independent person) of the finished task:

- The A3-7 engineer certifying the task must enter his name, CAA authorisation number, and full signature under "1st inspection".
- The independent second person must enter his name, CAA authorisation number or Pilots Licence number, and full signature under "2nd inspection".

This second person must be suitably qualified and may be:

- another A3-7 authorised engineer
- a qualified gyroplane pilot. In this case the pilot must append his Pilot's Licence number to his signature.

It is the second signatory's responsibility to ensure he/she understands the task and what it is they are inspecting and signing for.

Verification of Initials, Signature and Authorisation

The person performing the work must complete the "Tasks completed by" statement towards the end of the worksheet. The A3-7 authorised engineer must complete and sign the "Permit Maintenance Release" on the last page of the Worksheet.